uAvionix Corporation 300 Pine Needle Lane Bigfork, MT 59911 U.S.A.

FAA-APPROVED

AIRPLANE FLIGHT MANUAL SUPPLEMENT

for the

uAvionix AV-30-C

as installed on Piper PA-34-220T

Airplane Make and Model

Registration Number: Serial Number: 24

This supplement must be attached to the FAA-approved Airplane Flight Manual when the AV-30-C is installed in accordance with a follow on field approval to STC SA00410BO

The information contained herein supplements the basic manual only in those areas listed. For limitations, procedures, performance and loading information not contained in this supplement, consult the FAA-approved Airplane Flight Manual, markings, or placards.

DAVIDS

FAA Approved By: David S. Shanahan SHANAHAN

Allegheny FSDO EA03 101 Towne Square Way Suite 201 Brentwood, Pennsylvania 15102

Date: 04/29/2021

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US Department
of Transportation
Federal Aviation
Administration

OMB No. 2120-0020	Electronic Tracking Number
Exp: 01/31/2023	
	For FAA Use Only

US Departm	nent	MAJOR REPAIR AND ALTERATION							For FAA Use Only			
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date	work completed.)
	4/20/2021
Nationality and Registration Mark	Date
Completed installation of UAvionics AV-30-C Multi-Mode Indicator as a follow Type Certificate Number SA00410BO. The AV-30-C was installed in place of the coordinator and standby attitude indicator and functions as both. The installation accordance with UAvionics AV-30-C Installation Manual document UAV-10038 Electronic load analysis completed and found the maximum load draw is 1 Ame the equipment removed. A 2 amp curcuit breaker was installed per instruction balance and equipment list was revised to reflect changes. Completed GPS in previously installed Garmin GTN-750 using RS-232 serial input in accordance with installation instructions section 8.2.5. Transponder/ altimeter / encoder /and starequired by 14 CFR Parts 91.411 and 91.413 were completed. The installed Athe Garmin GTN-750, the aircraft main power source and the existing pitot and are no other systems that the AV-30 is interfaced to. Instructions for Continued be in compliance with UAviones AV-30-C Instructions for Continued Airworthin UAV-1004045-001 Rev. A dated 4/24/2020. Completed all system checks in a section 10 of installation manual. Completed compass swing and installed new card. All work previously desribed was entered into the aircraft's perminant recommunal supplement document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete document UAV-1004044-001 Rev. C to the aircraft flight in the complete docu	on to Supplemental he primary turn on was completed in 947-001 Rev. C. up, which is less than manual. Weight an terface from the with AV-30-C atic system tests AV-30 is interfaced to distatic system. There is Airworthiness will less document occordance with av compass deviation ords. Attached flight
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Additional Sheets Are Attached	